

# Full Halo Coronal Mass Ejections: Arrival at the Earth

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Abstract. A geomagnetic storm is mainly caused by a frontside coronal mass ejection (CME) hitting the Earth and then interacting with the magnetosphere. However, not all front-side CMEs can hit the Earth. Thus, which CMEs hit the Earth and when they do so are important issues in the study and forecasting of space weather. In our previous work (Shen et al., 2013), the de-projected parameters of the full-halo coronal mass ejections (FHCMEs) that occurred from 2007 March 1 to 2012 May 31 were estimated, and there are 39 front-side events could be fitted by the GCS model. In this work, we continue to study whether and when these front-side FHCMEs (FFHCMEs) hit the Earth. It is found that 59% of these FFHCMEs hit the Earth, and for central events, whose deviation angles  $\epsilon$ , which are the angles between the propagation direction and the Sun-Earth line, are smaller than 45 degrees, the fraction increases to 75%. After checking the deprojected angular widths of the CMEs, we found that all of the Earth-encountered CMEs satisfy a simple criterion that the angular width  $(\omega)$  is larger than twice the deviation angle ( $\epsilon$ ). This result suggests that some simple criteria can be used to forecast whether a CME could hit the Earth. Furthermore, for Earth-encountered CMEs, the transit time is found to be roughly anti-correlated with the de-projected velocity, but some events significantly deviate from the linearity. For CMEs with similar velocities, the differences of their transit times can be up to several days. Such deviation is further demonstrated to be mainly caused by the CME geometry and propagation direction, which are essential in the forecasting of CME arrival.

### 1 Introduction

The halo coronal mass ejections (CMEs), which appear to surround the occulting disk of coronagraphs, are preliminarily supposed to be propagating along the Sun-Earth line(*Howard et al.*, 1982). Under this assumption, the

front-side halo CMEs might be good candidates for Earthimpacted CMEs; however, not all of the front-side halo CMEs can hit the Earth. The ratio of the front-side halo CMEs hitting the Earth varied from 65% to 80%, which has been reported in different literature reports (Yermolaev and Yermolaev, 2006, and reference therein). In addition, most works are concerned about the geoeffectiveness of halo CMEs(e.g. Webb, 2002; Wang et al., 2002; Zhao and Webb, 2003; Zhang et al., 2007; Gopalswamy et al., 2007). The ratio of the front-side halo CMEs with geoeffectiveness varied from 45% to 71%. All of these works suggested that not all front-side halo CMEs can hit the Earth. Thus, what type of front-side halo CMEs can hit the Earth has been discussed by many authors(e.g. Wang et al., 2002; Zhang et al., 2003; Kim et al., 2005; Moon et al., 2005). Before the launch of the Solar Terrestrial Relations Observatory (STEREO), only coronagraph images and in situ measurements could be used to observe the CMEs near the Sun and the interplanetary CMEs (ICME) near the Earth respectively. Thus, direct connections between the CMEs near the Sun and the ICMEs near 1 AU might be unclear, especially during the solar maximum. Recently, using the large field of view observations from the Heliospheric imagers in the Sun-Earth-Connection Coronal and Heliospheric Investigation (SECCHI) (Howard et al., 2008) onboard STEREO, the propagation of CMEs could be well tracked continuously from the Sun to 1 AU. In this manner, the ejecta observed near the Earth and the CMEs that occurred near the Sun can be related in a more precise way. Can we re-investigate how many and what type of front-side halo CMEs can hit the Earth? In addition, we note that the apparent angular width threshold used to define the halo CMEs in previous works varied greatly, such as  $120^{\circ}$ ,  $130^{\circ}$ ,  $140^{\circ}$  and  $360^{\circ}$ . If we apply the apparent angular width =  $360^{\circ}$  only, will the ratio and the criteria of the front-side full halo CMEs arriving at the Earth be changed?

In addition, if a CME can hit the Earth, its arrival time becomes an important issue in space weather forecasting. Recently, various kinematics and magnetohydrodynamic (MHD) models have been developed to forecast the arrival time of CMEs(e.g. Fry et al., 2003; Odstrcil et al., 2004; Tóth et al., 2005; Mckenna-Lawlor et al., 2006; Feng and Zhao, 2006; Shen et al., 2007; Feng et al., 2007; Shen et al., 2010; Feng et al., 2010, and reference therein) In those models, the velocity, propagation direction and angular width of the CMEs are used as the initial parameters. However, the following basic questions are still not fully answered. What are the key parameters that determine the transit time of the CMEs from the Sun to the Earth? What is the extent of influence of the leading parameter? The CME's initial speed has been correlated with the transit time of the CME from the Sun to 1 AU(Cane et al., 2000; Wang et al., 2002; Zhang et al., 2003; Schwenn et al., 2005; Shanmugaraju and Vršnak, 2014). Some simple equations were established to calculate the possible arrival time of the CMEs based on their initial velocities. However, the deviation between the calculated transit times and the observations is large. One possible reason for this deviation is that the CME's velocity might change greatly during its propagation in the interplanetary space due to the influence of the background solar wind(e.g. Gopalswamy et al., 2000). Using a constant acceleration (or deceleration) assumption, Gopalswamy et al. (2001, 2005) developed an empirical CME arrival (ECA) model to predict the arrival time of Earth-directed CMEs. However, the acceleration (or deceleration) may also be changed during the propagation of CMEs in the interplanetary space. Recently, some other CME propagation time forecasting models were developed based on aerodynamic drag models(e.g. Vršnak et al., 2013). The aerodynamic drag models assume that the acceleration (or deceleration) of CMEs depends on the velocity difference between the CME and the background solar wind. Most of the above works are mainly focused on the velocity of the CME. Are there any other parameters that would exert significant influence on the propagation time of CMEs from the Sun to the Earth? How significant the influence of these parameters? The propagation direction of CMEs might be another important parameter. It has been taken into account in many CME arrival time forecasting models, such as the advanced version of the drag-based model (DBM, http://oh.geof.unizg.hr/DBM/dbm.php), the ENLIL model and other MHD models(e. g. Odstrcil, 2003). Recently, based on a self-similar expansion assumption and a theoretical computation (Davies et al., 2012), Möstl and Davies (2012) suggested that the propagation time of a CME is influenced by its propagation direction and the angular width.

In our previous work (Shen et al. (2013), referred to as Paper I hereafter), the projection effect of full-halo CMEs (FHCMEs) listed in the CDAW CME catalog(Yashiro et al., 2004) with an apparent angular width of  $360^{\circ}$  occurred from 2007 March 1 to 2012 May 31 was studied. In Paper I, the Graduated Cylindrical Shell (GCS) model (Thernisien et al., 2006, 2009; Thernisien, 2011) was applied on the STEREO/COR2 and SOHO observations to obtain the de-projected kinematic parameters of these FHCMEs. Among the total of 88 events studied in paper I, 48 events originated from the front of the solar disk. Table 1 in this paper and Table B in our online list (http://space.ustc.edu.cn/dreams/fhcmes/) present the parameters of these front-side full halo CMEs (FFHCMEs). Of the total of 48 FFHCMEs, there are nine events that could not be fitted by the GCS model. The remaining 39 events with well-established de-projected parameters will be studied in detail here. In this work, we will verify whether these FFHCMEs hit the Earth by using the continuous CME propagation observations of COR2, HI1, HI2 on board the STEREO spacecraft and the in situ measurements of the WIND and ACE satellites. Next, we attempt to answer the main questions of which and when the FFHCMEs will hit the Earth based on the de-projected parameters. In section 2, we introduce the method to determine the interplanetary counterpart for a given FFHCME. Based on the list of the FFHCMEs and their associated interplanetary counterparts, the type of the Earth-encountered FFHCMEs will be discussed in section 3. In section 4, the parameters that affect the transit time of CMEs from the Sun to the Earth will be discussed. A conclusion and some discussions of the results will be provided in the last section.

# 2 Methods

The interplanetary magnetic field and the solar wind plasma observations from WIND and ACE satellites are used to examine whether an interplanetary CME (ICME) was recorded near the Earth the six days following the launch of the CME with an Earthward potential. Previous works used different criteria to identify the ICME(e. g. Wang et al., 2002, 2004; Jian et al., 2006, and reference there in). In this paper, the following characteristics are used in the investigation: (1) enhanced magnetic field intensity, (2) smoothly changing field direction, (3)relatively low proton temperature, (4) low proton plasma beta, and (5) bidirectional streaming of electrons. An ICME structure is recognized when it fits at least three of the criteria listed above. The detailed observations of the identified ICME are provided in our online website (http://space.ustc.edu.cn/dreams/fhcmes/). Figure 1 shows the in situ observations from 2010 April 3 11:00UT to 2010 April 9 11:00UT, in the next six days after the launch of the CME occurred at 2010 April 3 10:33:58UT. For this event, an obvious ICME was recorded from 2010 April 5 $12{:}00\mathrm{UT}$  to 2010 April 6 16:00UT by WIND and ACE, which is indicated by the gray region in the Figure 1. This ICME is treated as the possible interplanetary counterpart of the 2010 April 3 10:33:58UT CME. In addition, a shock ahead of this ICME impacted the Earth at approximately 2010 April 5 08:00UT.

If there is at least one ICME recorded in the next six days after a CME was launched, the Time-Elongation Angle maps (J-maps) (e. g. Sheeley et al., 1999; Davis et al., 2009) are used to perform further verification of the association between the ICMEs recorded near the Earth and the FFHCMEs near the Sun. A 64-pixel-wide slice is placed along the ecliptical plane in the running-difference images from COR2, HI1 and HI2 on board STEREO, and the slices adopted at different times are stacked to obtain the J-map. A 64-pixel corresponds to  $\sim 0.95$  solar radius for COR2 and  ${\sim}1.25^\circ$  and  ${\sim}4.38^\circ$  of the elonagtion angle in the HI1 and HI2 field of view respectively. Figure 2 shows the J-map for the 2010 April 3 10:33:58UT CME. As seen in Figure 2, after the CME take-off (the vertical solid line), a black-white track that corresponds to the front of this CME extended to the region with an elongation angle of  $\sim 60^{\circ}$ . At the time that the ICME was observed at 1 AU (indicated by the vertical dashed line), the front of this CME also reached a location near the Earth (shown by the horizontal dashed-dotted line). Thus, the ICME observed near the Earth from 2010 April 5 12:00UT to 2010 April 6 16:00UT was well associated with the 2010 April 3 10:33:58UT CME. In this manner, we found that a total of 27 of the 48 FFHCMEs considered hit the Earth. The 7th column of Table 1 shows the arrival time of the shock driven by the CME. The 8th and 9th columns present the beginning and ending times of the associated

	CME data	Direction	6	( 1	Varan	$T_{\pi\pi}$ .	Trans bogin	Transa and
1	2009/12/16 04.30.03	E07 N09	е 11	$\frac{\omega}{45}$	<u>VGCS</u> 	1 Shock	2000/12/10 00.40	2000/12/20 00.22
2	2000/12/10 01.00.00	E06 S07	0	81	/81	2010/02/11 00:00	2000/12/10 00.10 2010/02/11 13.00	2000/12/20000.22
2	2010/02/07 03.04.03 2010/02/12 13.42.04	E00,507	11	8/	550	$2010/02/11\ 00.00$ $2010/02/15\ 17.40$	2010/02/11 15.00 2010/02/16 04.00	$2010/02/11\ 22.00$ $2010/02/16\ 12.00$
	2010/02/12 10.42.04 2010/04/03 10.33.58	E01,101	$\frac{11}{97}$	8/	853	2010/02/10 11.40 2010/04/05 07.56	2010/02/10 04.00 2010/04/05 12.00	2010/02/10 12.00
5	2010/05/23 18.06.05	W16 N07	17	70	365	2010/04/05 07.50 2010/05/28 01.58	2010/04/05 12.00 2010/05/28 10.00	2010/04/00 10.00 2010/05/20 17.00
6	2010/05/23 10.00.05	W26 S06	26	63	552	2010/03/28 01.58	2010/03/28 13.00	2010/03/23 11.00
	2010/05/24 14.00.05 2010/08/01 12.42.05	F28 N20	42	03	1969		2010/08/04 10:00	2010/08/05 02:00
8	2010/08/01 13.42.05 2010/08/07 18.26.06	E36,1120	36	90	770	2010/08/03 17.00	2010/08/04 10.00 2010/08/11 05:00	2010/08/03 02.00 2010/08/12 17.00
	2010/08/07 18.30.00 2010/08/14 10.12.05	W42 S11	42	110	864		2010/08/11 05.00	2010/08/12 17.00
9	2010/08/14 10.12.03 2010/12/14 15.26.05	W25 N20	40	119	004			
11	2010/12/14 15.50.05 2011/02/14 18.24.05	W08 N01	30	61	365		2011/02/18 10.20	2011/02/18 10:30
10	2011/02/14 10.24.05	W05,N01	8	140	764		2011/02/18 10.30	2011/02/10 19.30 2011/02/20 08.00
12	2011/02/13 02.24.05 2011/02/07 20:00:05	W34 N33	45	140	1022	2011/02/18 01.00	2011/02/18 20.00	2011/02/20 08.00
13	2011/05/07 20.00.05 2011/06/02 08.12.06	F30 S03	20	02	061	2011/06/04 20:00	2011/06/05 02:00	2011/06/05 18:00
14	2011/00/02 08.12.00	E30,505	- 30	92	901	2011/00/04 20:00	2011/00/05 02:00	2011/00/00 18:00
10	2011/00/07 00.49.12	F20 N07		02	064		2011/06/22 06:00	
10	2011/00/21 03.10.10	W10 N12	15	90	904	2011/00/23 $02.00$	2011/00/23 00:00	2011/00/24 00.00 2011/08/05 17.20
10	2011/08/03 14:00:07	W10, N12	10	124	925	$2011/08/04\ 21.13$	2011/08/05 05.50	2011/08/05 17.50 2011/08/07 22.00
10	2011/08/04 04:12:03	W 30, N24	42	107	1504	2011/08/05 17:50	2011/08/00 22:00	2011/08/07 22:00
19	2011/08/09/08:12:00	W45,N10	47	100	1394			
20	2011/09/00 02:24:03		4.4	116			2011/09/08 10:00	2011/09/09 12:00 2011/00/10 15:00
21	2011/09/00 25:05:57	W41,119	44	110	901	2011/09/09 12:00	2011/09/10 05:00	2011/09/10 15:00
22	2011/09/22 10:48:00	$E_{12}$ , NOC	12	131	1020			
20	2011/09/24 12:48:07 2011/00/24 10:26:06	E47,N00	47	119	1708	2011/09/20 12:00	2011/09/20 20:00	2011/09/28 00:00
24	2011/09/24 19:50:00 2011/10/22 01:25:52							
20	2011/10/22 01:23:35							
20	2011/10/22 10:24:05	 E49 N96	10	E1		2011/10/24 17:58	2011/10/23 00:00	2011/10/20 10:00
21	2011/10/27 12:00:00	E42,N20	40	01 170	1074			
20	2011/11/09 15:50:05	$E_{30,1NZ4}$ W25 N17	4Z 20	177	1074	2011/11/12 00:20	2011/11/12 14:01 2011/11/20 00.12	2011/11/15 11:09 2011/11/20 04:52
29	2011/11/2007:12:00	W 55,IN17	90	111	900	$2011/11/26\ 20:31$	$2011/11/29 \ 00:12$	2011/11/29 04:55 2012/01/06 02:42
30	2012/01/02 13:12:40 2012/01/16 02:12:10		C A	194	050	2012/01/05 15:50	2012/01/04 00:00	2012/01/00 02:45
31	2012/01/10 $05:12:10$	E07,N09 E17 N42	45	124	900			
32	2012/01/19 14:50:05	E17,1N45	40	141	1090	2012/01/22/03:10	2012/01/23 00:13	2012/01/24 15:09
24	2012/01/25 04:00:05	W10,N41	40	195	1022	2012/01/24 14:50		
34	2012/01/20 04:30:03	W71,N30	79	00	1033			
30	2012/01/27 18:27:32 2012/02/00 21:17:26	E42 N20	19	70	649	2012/01/30 15:50		
30	2012/02/09 21:17:50 2012/02/10 20:00:05	E42,N29	49	79	040 E02			
31	2012/02/10 20:00:03	$E_{20}$ , $N_{20}$	01 64	195	000		2012/02/14 17:38	2012/02/10 00:55
30	2012/02/23 08:12:00	$F_{41} N_{27}$	04	150	442	2012/02/20 20:38	2012/02/27 17:35	2012/02/28 15:40
39	2012/03/04 11:00:07 2012/02/05 04:00:05	E41,N27	47	100	1190			
40	2012/03/05 04:00:05		17	140		2012/03/07/03:28	2012/03/07 20:30	2012/03/08 11:41 2012/02/11 08:02
41	2012/03/07 00:24:00 2012/02/00 04:26:00	100,1003	41	140	2012	2012/03/08 10:34	2012/03/09/05:19	2012/03/11 08:03
42	2012/03/09/04:20:09	W16 N10	0	107	1100	2012/02/12 00.17		2012/02/15 00.40
40	2012/03/10 16:12:00 2012/02/12 17:26:05	W27 N22	20 47	107	1525	2012/03/12/08:17 2012/02/15/12.05	$\frac{2012}{09} \frac{12}{12} \frac{21:41}{2019}$	2012/00/10/08:42 2010/00/16/10.00
44	2012/03/13 1/:30:05 2012/04/05 21.25.07	VV 37,1V33	41	104	1929	2012/03/13 12:05	2012/03/10 00:51	2012/03/10 12:09
40	2012/04/05 21:25:07		41	0.4	<u> </u>			
40	2012/04/09 12:30:07	VV 40, IN 12	41 96	94	092			
41	2012/05/12/00:00:05	E20,510	20	60	939			
4ð	2012/00/17 01:48:05							

Table 1: The GCS model's parameters and the times of the associated ICMEs of the FFHCMEs occurred from 1 March 2007 to 2012 May 31



Figure 1: The WIND and ACE observations from 2010 April 3 11:00UT to 2010 April 9 11:00UT. From top to the bottom, they are the magnetic field strength (|B|) from WIND, the elevation ( $\theta$ ) and azimuthal ( $\phi$ ) of field direction based on WIND observations, the ditribution of the electron intensity at different position angle from ACE, solar wind speed ( $V_{SW}$ ) from WIND, proton density ( $N_p$ ) from WIND, proton temperature ( $T_p$ ) from WIND and the ratio of proton thermal pressure to magnetic pressure ( $\beta_p$ ) calculated based on the WIND observations.



Figure 2: The time-elonagation (J-map) from 2010 April 3 to 2010 April 7. The solid vertical line shows the time of the CME butst. The dashed line shows the time of the ICME recorded at 1AU. The horizotal dashed-dotted line shows the elongation angle of the Earth.



Figure 3: The distribution of the propagation directions of these FFHCMEs. The solid dots show the Earthencountered FHCME events while the open circles show the events which did not hit the Earth. The diamonds show the events in which only shocks driven by the CMEs hit the Earth.

ICMEs. The '—' symbol denotes that no ICME or shock was associated with this FFHCME.

### **3** Which CMEs arrived at the Earth?

Figure 3 shows the distribution of the propagation directions of all 39 FFHCMEs whose de-projected parameters were well established. It should be noted that, in the following analysis, only these 39 FFHCMEs were used. Those propagation directions were distributed in a large range, from E73° to W71° of longitude. According to their propagation longitudes, all 39 FFHCMEs can be classified into 20 eastern events and 19 western events. However, the propagation directions distribution in the north and south sides exhibit an obvious asymmetry. There are a total of 30 events propagating in the northern heliosphere and only nine events in the southern heliosphere. This asymmetry might be caused by the fact that the northern hemisphere is more active than the southern hemisphere in the ascending phase of the 24th solar cycle(e.g. Svalgaard and Kamide, 2013).

In the 39 events whose de-projected parameters have been obtained, 59% (23) of them hit the Earth. The solid dots in Figure 3 show the events that arrived at the Earth. The figure shows that their propagation longitudes are distributed in a range of [E47°, W61°], which is more narrow than the longitudinal distributions of all FFHCMEs. Note that the most eastern of the Earth-encountered events propagated at E47°, and the most western event came from W61°. This observation is consistent with the previous result that it is difficult for the east limb CMEs to hit the Earth(*Wang et al.*, 2002; *Zhang et al.*, 2003). In addition, the longitude range of most (21/23) of the Earth-encountered FFHCMEs is [E40°,W40°]. Meanwhile, for the FFHCMEs whose propagating longitudes located in this region, approximately 72% of them arrived at the Earth. This result suggests that the



Figure 4: The angular width  $\omega$  varied with  $\epsilon$  for the FFHCMEs. The solid dots show the Earth-encountered CMEs while the open circles show the not Earth-encountered CMEs. The diamonds show the events in which only shocks driven by the CMEs hit the Earth.



Figure 5: The sketch map of the CME's self-similar expansion model.

central events with propagation longitudes in the range of  $[E40^{\circ}, W40^{\circ}]$  are more likely to hit the Earth.

The deviation angle  $\epsilon$ , which is defined as the angle between the propagation direction and the Sun-Earth line, is used to discriminate among the possible Earth-encountered CMEs. From Figure 4, 24 FFHCMEs propagated with  $\epsilon \leq 45^\circ$  and 75% (18) of them hit the Earth. Meanwhile, for the 13 FFHCMEs with  $\epsilon \leq$  30°, 77% (10) of them arrived at the Earth. For comparison, only one main body or the flux rope structure of five limb FFHCMEs with  $\epsilon > 60^{\circ}$  hit the Earth. It should be noted that, one shock driven by another limb CME hit the Earth. But, in this work, we mainly discussed whether the main body or the flux rope like structure of the CME hit the Earth. This observation confirms the previous result that the central CMEs can hit the Earth with higher possibility. In addition, five events with  $\epsilon > 45^{\circ}$ arrived at the Earth. Upon checking their de-projection parameters, we find that all of these events are wide events and that their minimum angular width is  $103^{\circ}$ . This observation indicates that the limb CMEs can also impact the Earth if they are wide.

Simply by assuming that a CME moves as a self-similar expansion ball (e.g. *Davies et al.*, 2012; *Möstl and Davies*, 2012) as shown in Figure 4, one can expect that the CME can hit the Earth when its angular width is larger than twice

the deviation angle, i.e.,  $\omega > 2\epsilon$ . The dotted-dashed line in Figure 5 indicates  $\omega = 2\epsilon$ . Based on the previous analysis, only CMEs located in the region above this line could hit the Earth. The observations that all of the Earth-encountered FFHCMEs are located in the upper region confirm the above conclusion. In addition, a large fraction (74%, 25/34) of FFHCMEs that fit the condition of  $\omega > 2\epsilon$  hit the Earth. For comparison, all four events under the dotted-dashed line did not hit the Earth. This result indicates that the  $\omega > 2\epsilon$  can be a useful criterion to forecast whether a CME would hit the Earth; therefore, the angular width is another important parameter in the space weather forecasting model.

Note that some CME events that were launched from regions close to the solar-disc center or fit the criterion of  $\omega > 2\epsilon$  did not hit the Earth. One possible reason is that those CMEs may be deflected during their interaction with other CMEs(e.g. Xiong et al., 2009; Wang et al., 2011; Lugaz et al., 2012; Shen et al., 2012). Lugaz et al. (2012) studied the interaction between the CMEs occurred on 2010 May 23 and 2010 May 24, which are the No 5 and No 6 events, respectively, in our list. They found that the Earth-direct CME on 2010 May 24 had missed the Earth, which was a result of the interaction with the 2010 May 23 CME. In our investigation, six events with  $\epsilon < 45^{\circ}$  and  $\omega > 2\epsilon$  did not hit the Earth. By carefully checking the heliospheric images from STEREO HI1 and HI2, we found that two events might be affected by their interaction with other CMEs: the 2012 May 24 event, which was studied by Lugaz et al. (2012), and the 2012 March 9 event. Why did the CMEs of the other four events not hit the Earth? A possible reason is that the CMEs might be deflected during their propagation in the corona (e.g. MacQueen et al., 1986; Gopalswamy and Thompson, 2000; Cremades and Bothmer, 2004; Kilpua et al., 2009; Gui et al., 2011; Shen et al., 2011; Zhou and Feng, 2013) as well as in the interplanetary space (e.g. Wang et al., 2004, 2006, 2013).

### 4 When did CMEs arrive at the Earth?

Figure 6 shows that the transit times of the CMEs from the Sun to 1 AU varied with the de-projection velocities of the CMEs. In Figure 6, the propagation time exhibits an anti-correlation with the de-projected velocities, but the dispersion is large. For the CMEs with similar velocities, the difference of the transit time can be up to tens of hours. For example, there are eight events whose velocities are  $\approx 1000 km.s^{-1}$  (from  $800 km.s^{-1}$  to  $1200 km.s^{-1}$ ). The transit time of these events from the Sun to 1 AU varied from 37 hours for the 2011 August 3 event to 82 hours for the 2012 January 19 event. The difference between the propagation times is approximately 2 days.

Why do these CMEs with similar velocities have quite a different transit time from the Sun to 1 AU? It is probably because the CMEs have a circular-like front, and it is not always true that the leading front of a CME encounters the Earth. This effect means that in the case of 'noncentral' impact, the CME forehead reaches a distance larger than 1 AU at the time when the CME arrival is recorded in the in situ observations near the Earth. A well-investigated case can be found in the most recent work by *Wang et al.* (2013), in which a CME occurred on 2008 September 13 passed through both the WIND and STEREO-B spacecraft



Figure 6: The propagation times varied with the  $v_{GCS}$  of the Earth-encountered FFHCMEs. The rectangle include the eight events with similar velocity of  $1000\pm200$  km.s<sup>-1</sup>.

at 1 AU. The arrival time of this CME at STEREO-B was approximately two days later than that at WIND. Again, we assume that the CME is a self-similar expansion ball radially propagating along the direction with a deviation angle of  $\epsilon$ . When the observatory at 1 AU detected this CME, the real propagation distance of the CME tip along its propagation direction  $(D_{rf}$ , the length of SC in Figure 5) was obviously larger than 1 AU  $(D_{AU})$ . The difference between the  $D_{rf}$ and 1 AU,  $\Delta D = D_{rf} - 1AU$ , depends on the angular width of the CME  $(\omega)$  and the propagation direction  $\epsilon$ . The  $D_{rf}$ could be obtained by:

$$D_{rf} = R_{CME} + \frac{R_{CME}}{\sin(\frac{\omega}{2})} \tag{1}$$

in which  $R_{CME}$  is the radius of this CME, which can be calculated from the equation of:

$$\left(\frac{R_{CME}}{\sin(\frac{\omega}{2})}\right)^2 + D_{AU}^2 - 2D_{AU}\frac{R_{CME}}{\sin(\frac{\omega}{2})}\cos\epsilon = R_{CME}^2 \tag{2}$$

From Equations 1 and 2, the real propagation distance  $D_{rf}$  can be determined if the angular width  $\omega$  and deviation angle  $\epsilon$  are all known. To improve the comparison, we choose eight events with similar velocities in the range of  $1000\pm200 km.s^{-1}$  for the further study. Figure 7 shows that calculated real propagation distance  $D_{rf}$  varied with the observed propagation time of these eight events based on the de-projected parameters obtained in paper I. Additionally, the  $D_{rf}$  of these events varied over a large range, from 1.04 AU to 1.53 AU. As shown in the previous analysis,  $D_{rf}$ could be treated as the real propagation distance of these CMEs along their propagation direction when they eventually arrived at 1 AU. As shown in Figure 7, the propagation time and the real propagating distance of these events have obvious positive correlation. This result indicates that the different transit time of CMEs with similar velocities might be caused primarily by the different part of the circular-like CME front arriving at 1 AU. From equations 1 and 2, the propagation distance of the CME tip is related to the angular width and the propagation direction. Thus, the true



Figure 7: The CME's real propagation distances  $D_{rf}$  varied with propagation times for the Earth-encountered FFHCMEs with  $800km.s^{-1} \leq v_{gcs} \leq 1200km.s^{-1}$ . The real propagation distances  $D_{rf}$  are calculated based on the SSE model and the propagation time is obtained from the real obsrvations.

angular width and the propagation direction are all important parameters in the CME arrival time forecasting as well as the CME's velocity and the background solar wind speed (e.g. Vršnak and Žic, 2007; Temmer et al., 2011).

The propagation of CMEs in the interplanetary space can be described by an aerodynamic drag model(e.g. *Chen*, 1996; *Maloney and Gallagher*, 2010; *Vršnak et al.*, 2013; *Lugaz and Kintner*, 2012, and reference therein). Here, we use a simplified equation of the aerodynamic drag model from *Maloney and Gallagher* (2010) as:

$$\frac{dv_{CME}}{dr} = -Cr^{-1/2}(v_{CME} - v_{SWE})^2$$
(3)

in which C is a constant number. The dashed line in Figure 7 shows the result of the aerodynamic drag model by assuming the initiation speed of a CME is  $1000 km.s^{-1}$ , the solar wind speed  $v_{SWE}$  is 450  $km.s^{-1}$ . The value  $C = 1.35 \times 10^4$  is obtained from a fitting process. In this figure, almost all of the points are close to the dashed line. Therefore, the propagation process of these CMEs could be well described by the aerodynamic drag model. Thus, the self-similar expansion model combined with the aerodynamic drag model might be a powerful tool to forecast the Earth arrival times of CMEs.

### 5 Conclusion and Discussion

In this work, we studied whether and when the frontside full halo CMEs that occurred from 2007 March 1 to 2012 May 31, hit the Earth. The de-projected parameters of those CMEs were obtained in our previous work(*Shen et al.*, 2013). The in situ observations combined with the SEC-CHI/COR2 SECCHI/HI-1 and SECCHI/HI-2 observations from STEREO were used to verify whether these FFHCMEs hit the Earth. We conclude the following:

1. Approximately 59% of the FFHCMEs studied in this work arrived at the Earth. The central CMEs, which propagated in the longitudinal range [E40°, W40°] or  $\epsilon \leq 45^{\circ}$ , can arrive at the Earth with higher probability.

- 2. The FFHCMEs with an angular width  $\omega$  of more than twice the deviation angle  $\epsilon$  can hit the Earth. All of the Earth-encountered events fit the criterion  $\omega > 2\epsilon$ , and 74% of the FFHCMEs events that fit the criterion of  $\omega > 2\epsilon$  hit the Earth. Thus, the simple criterion  $(\omega > 2\epsilon)$  might be a useful tool to forecast whether a CME will hit the Earth.
- 3. The propagation times exhibit an overall anticorrelation with the de-projected velocities. The selfsimilar expansion model can be used to adequately explain the different transit time of the CMEs from the Sun to 1 AU with similar velocities. Furthermore, we suggest that the self-similar expansion model combined with the aerodynamic model is a simple and useful tool to forecast the arrival time of CME.

Based on the previous analysis, we found that the CME's real propagating distance,  $D_{rf}$ , is determined by its angular width and the propagation direction. The propagating distance of the CME tip might be larger than 1 AU when the flank of the CME hit the Earth. To evaluate the influence of this effect, the values of  $D_{rf}$  with different angular widths  $\omega$  and deviation angles  $\epsilon$  were calculated. Assuming the deviation angle  $\epsilon$  varied from  $1^{\circ}$  to  $90^{\circ}$  and the angular width  $\omega$  varied from 1° to 180°, Figure 8 shows the distribution of the  $D_{rf}$  values for the different cases. Note that the CMEs in the lower right half of the plot ( $\omega < 2\epsilon$ , white regions in Figure 8) can not hit the Earth based on the SSE model. In other cases with  $\omega > 2\epsilon$ ,  $D_{rf}$  varied over a wide range, from  $\approx 1$  AU to more than 60 AU. Considering that a 10% uncertainty in the forecasting of the CME arrival time is acceptable, the influence of the different propagation direction and angular width on the travel time of a CME should be carefully considered in the cases with  $D_{rf} > 1.1$  AU. In a large fraction (60%) of the cases we discussed, the  $D_{rf}$  are larger than 1.1 AU, and those parameters must be considered. Inspecting Figure 8, one finds that at a given fixed value of  $\epsilon$ ,  $D_{rf}$  increases as the angular width decreases. Thus, the  $D_{rf}$  for narrower CMEs should be calculated first to forecast the arrival time of the CMEs. In addition,  $D_{rf}$  varies greatly with the change of the deviation angle, as shown in Figure 8. Larger  $\epsilon$  values correspond to larger  $D_{rf}$  values. Figure 9 shows the maximum and minimum values of  $D_{rf}$ as a function of the deviation angle  $\epsilon$  by assuming that the CME angular width varies from  $1^{\circ}$  to  $180^{\circ}$ . It is found that both values increase with increasing  $\epsilon$ . Particularly, when  $\epsilon > 25^{\circ}$ , the values of  $D_{rf}$  are always larger than 1.1 AU. Thus, combined with the previous results, we suggest that for narrow CMEs or the CME propagated with  $\epsilon > 25^{\circ}$ , the influence of the propagation direction and angular width on the CME's transit distance is large, and the  $D_{rf}$  must be carefully calculated in the CME arrival time forecasting.

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Figure 8: The dependence of  $D_{rf}$  on different combinations of the angular width  $\omega$  and deviation angle  $\epsilon$ .



Figure 9: The maximum and minimum values of  $D_{rf}$  as a function of the deviation angle  $\epsilon$ . The horizontal dashed line shows the  $D_r = 1.1$  AU.

satellites. The dataset we used are SOHO LASCO/C2, SOHO LASCO/C3, STEREO COR2, STEREO H11, STEREO H12, WIND MFI, WIND SWE and ACE SWEPAM. This work is supported by the Chinese Academy of Sciences (KZZD-EW-01), grants from the 973 key project 2011CB811403, NSFC 41131065, 41274173, 40874075, 41121003, and 41304145, CAS the 100-talent program, KZCX2-YW-QN511 and startup fund, and MOEC 20113402110001, the fundamental research funds for the central universities (WK208000031) and the Strategic Priority Research Program on Space Science, the Chinese Academy of Sciences (Grant No. XDA04060801).

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