

# In-flight calibration of the magnetometer on the Mars orbiter of Tianwen-1

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Received February 5, 2023; accepted April 14, 2023; published online July 4, 2023

Mars orbiter magnetometer (MOMAG) is one of seven science payloads onboard Tianwen-1's orbiter. Unlike most of the satellites, Tianwen-1's orbiter is not magnetically cleaned, and the boom where placed the magnetometer's sensors is not long enough. These pose many challenges to the magnetic field data processing. In this study, we introduce the in-flight calibration process of the Tianwen-1/MOMAG. The magnetic interference including spacecraft generated dynamic field and the slowly-changing zero offsets are cleaned in sequence. Then the calibrated magnetic field data are compared with the data from the Mars atmosphere and volatile EvolutionN (MAVEN). We find that some physical structures in the solar wind are consistent between the two data sets, and the distributions of the magnetic field strength in the solar wind are very similar. These results suggest that the in-flight calibration of the MOMAG is successful and the MOMAG provides reliable data for scientific research.

**magnetometer, in-flight calibration, Martian magnetic field, Tianwen-1**

**Citation:** Zou Z X, Wang Y M, Zhang T L, et al. In-flight calibration of the magnetometer on the Mars orbiter of Tianwen-1. *Sci China Tech Sci*, 2023, 66: 2396–2405. <https://doi.org/10.1007/s11431-023-2401-2>

## 1 Introduction

Tianwen-1, the first China's Mars exploration mission, contains an orbiter running on a large elliptical polar orbit [1]. It can arrive 2.8–4.3  $R_M$  ( $R_M$  refers to radius of Mars) to observe the farther magnetotail region [2], filling the gap of the existing Mars explorations, e.g., Mars global surveyor

(MGS) [3] and the Mars atmosphere and volatile EvolutionN (MAVEN) [4]. The Mars orbiter magnetometer (MOMAG) is designed to observe the magnetic field around Mars and has three main goals with cooperation with other instruments: (1) to explore the *in-situ* environment of the Martian ionosphere, induced magnetosphere and interplanetary space; (2) to study the interaction mechanism between these regions; (3) to study the ionospheric conductivity and currents together with the measurements of Mars rover

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magnetometer [1,5].

Most magnetometers are placed on the top of a long boom to keep far away from the spacecraft which may have significant interference on the measured magnetic field, or are required a strict magnetic cleanliness programme [6–9]. Venus Express is an exception, of which the spacecraft was not magnetically cleaned, but the magnetometer on board the spacecraft successfully provided excellent data after the careful in-flight calibration [10]. Tianwen-1 orbiter did not equip the magnetic cleanliness program due to budget. The measured magnetic field data could be severely affected by operations of the instruments onboard the orbiter, solar array driving mechanism changes, antenna transmission effects, reaction wheel effects, and slow thermal drift [11], which bring a challenge for data processing. Thus, the in-flight calibration becomes very important for MOMAG to provide reliable scientific data.

The magnetic field in the solar wind has some natural physical characteristics, which could be used in the magnetometer calibration. In the early stage, researchers found that over some solar rotations the averages of the magnetic field components are zero, and the method could only get an offset for several months consequently [12]. Later, some calibration methods based on the Alfvénic fluctuation, which occurs frequently in the solar wind and during which the magnetic field strength remains almost unchanged, were developed [12–16]. Davis-Smith method is a classical method based on the assumption that the variance of the squared magnetic field magnitude is minimum [15]. Other methods have different assumptions such as that the changes in the field magnitude and the changes in the inclination of the field to any one of the three coordinate axes have no correlation [13,16], and the fluctuations are transverse to the ambient field [16].

Afterwards, some methods based on mirror mode structures and current sheets were also developed for the zero offset calibration [17–20]. They assumed that the magnetic field is parallel or perpendicular to the maximum variance direction, respectively, according to the minimum variance analysis (MVA) [21]. Although the practicality of these methods still needs validation, they potentially may be complementary for the in-flight calibration in magnetosheath or magnetosphere where clear Alfvénic fluctuations appear rarely.

In the first several months of MOMAG routine operation, the orbital period is about 7.8 h with more than 50% of time in the solar wind. Thus, we can use the properties of Alfvénic fluctuation to calibrate the MOMAG data. Here, the recently developed Wang-Pan method [14] is used. In the next section, we first introduce the clean method of the spacecraft generated dynamic field due to the operation of instruments. In Sect. 3, we introduce the calibration of the MOMAG's zero offset. Sect. 4 shows the processed data and compares

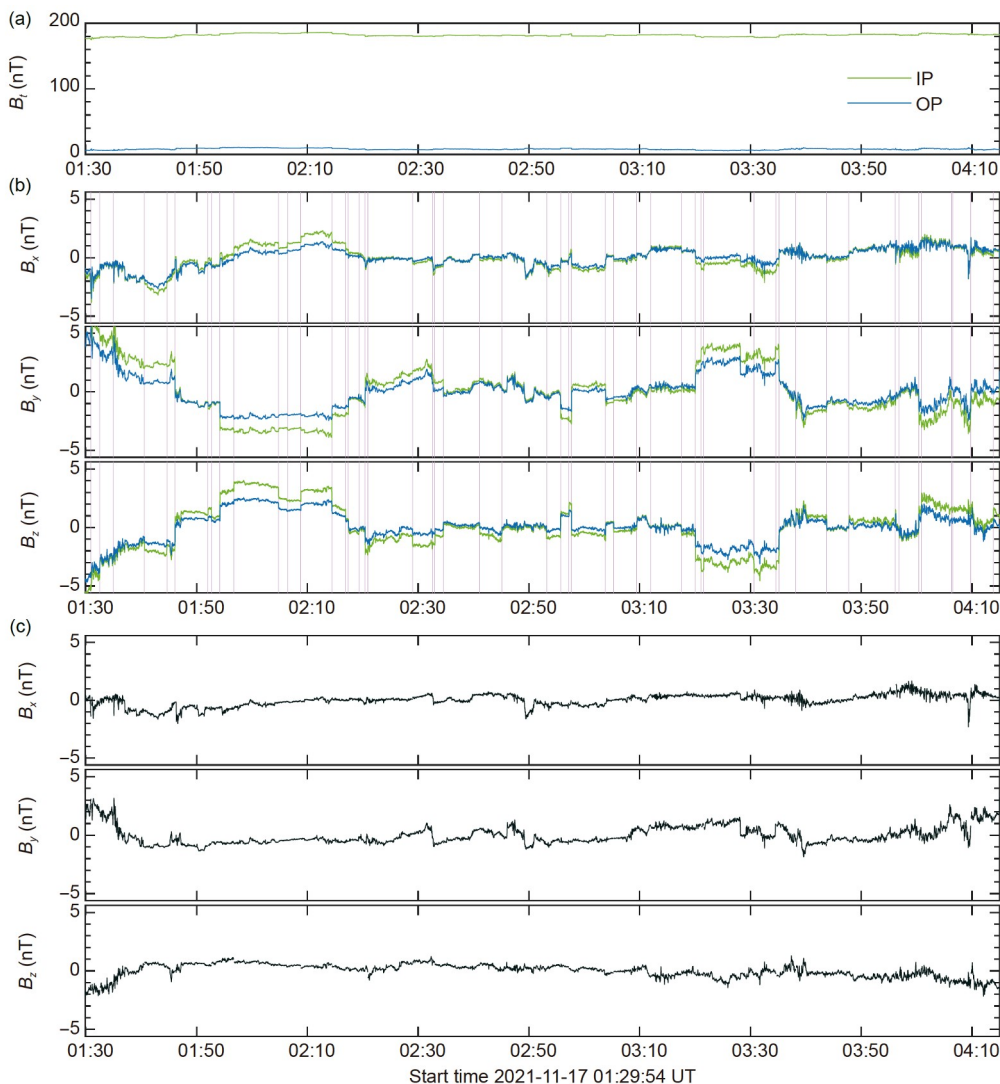
them with the calibrated magnetic field data obtained by the MAVEN spacecraft. Sect. 5 briefly summarizes the main results of this paper.

## 2 Clean method for dynamic field

MOMAG has two sensors, one is mounted at the top of the boom (outer sensor), which is 3.19 m away from the orbiter surface, and the other is at the middle of the boom (inner sensor), 2.29 m away from the orbiter surface [1,5]. Both of them measure the magnetic field independently, which are severely interfered by spacecraft operations. Such interferences contain two types. One appears like an artificial jump in the magnitudes of magnetic field components due to the operations, such as turning on/off, of the instruments onboard the orbiter. It frequently occurs with a rate of hundreds of times a day. The other type is a kind of systematic offset including the interferences from the persistent currents in spacecraft and the offset of the sensors themselves. It does not change too much over hours to days or even weeks.

Ness et al. [22] proposed a theoretic method, called gradiometer technique, to remove spacecraft interference by dual sensors. In their work, the interference magnetic field can be approximated by spherical harmonic coefficient expansions. The simplest assumption is the dipole field approximation, and Rong et al. [23] offered a technique to diagnose the dipole source from the measured magnetic field data. However, in practice, what we need is the fact that the spacecraft magnetic field decreases with the distance, i.e., the interference acting on the outer sensor is much weaker than that on the inner sensor. And more importantly, based on the equations of the spherical harmonic expansions in Ness et al.'s paper [22], the ratio of interference fields acting on the two sensors should be some fixed values as long as the interference sources do not change randomly, and the dipole field assumption is not required in principle. Figure 1(a) shows the uncalibrated magnetic field strength at the two sensors between 01:30 and 04:15 UT on 17 November 2021. The magnetic field at the inner sensor is more than one order higher than that at the outer sensor. Therefore, following the above thoughts, we start to process the data.

For the first type of the interferences, i.e., the artificial jumps, the magnetic field components change their values within 1 s with different magnitudes at the two sensors. Thus, we can easily recognize them and distinguish them from real jumps, which have the same magnitude at both sensors, in the solar wind. As an example, some artificial jumps can be clearly seen in Figure 1(b), in which the three components of the uncalibrated magnetic field detected by the two sensors are plotted. For a clear view, we subtract their mean and translate them to near zero. Note that the zero offset is not cleaned yet, and therefore the values of the



**Figure 1** (a) The magnetic field strength detected by the outer probe to the inner probe. (b) The three components of the original magnetic field detected by the outer probe (blue) and the inner probe (green) in the spacecraft coordinates between 01:30 and 04:15 UT on 17 November 2021. The artificial jumps are marked in purple lines. (c) The magnetic field with the artificial jumps removed (black) in the outer probe. The three components are presented in the spacecraft coordinates.

magnetic field scaled on the left  $y$ -axis in Figure 1(b) are meaningless for the scientific research.

Some of these jumps are physical but more are artificial, and the artificial jumps are indicated by purple lines. The artificial jumps at inner sensor are larger than those at outer sensor. Then we get rid of these artificial jumps by using the algorithm that was successfully applied on the magnetometer of Venus Express [11,24] (refer to the paper for details). After cleaning the artificial jumps, we could find that the magnetic field components become much more stable than original data, as shown in Figure 1(c).

### 3 Calibration method of the zero offset

After removing the dynamic fields, only the slowly-changing

zero offset is unremoved in the data. We use the Alfvénic fluctuation events in the solar wind to calibrate it based on the Wang-Pan method [25]. We first develop a semi-automated procedure to choose highly Alfvénic fluctuation events as follows.

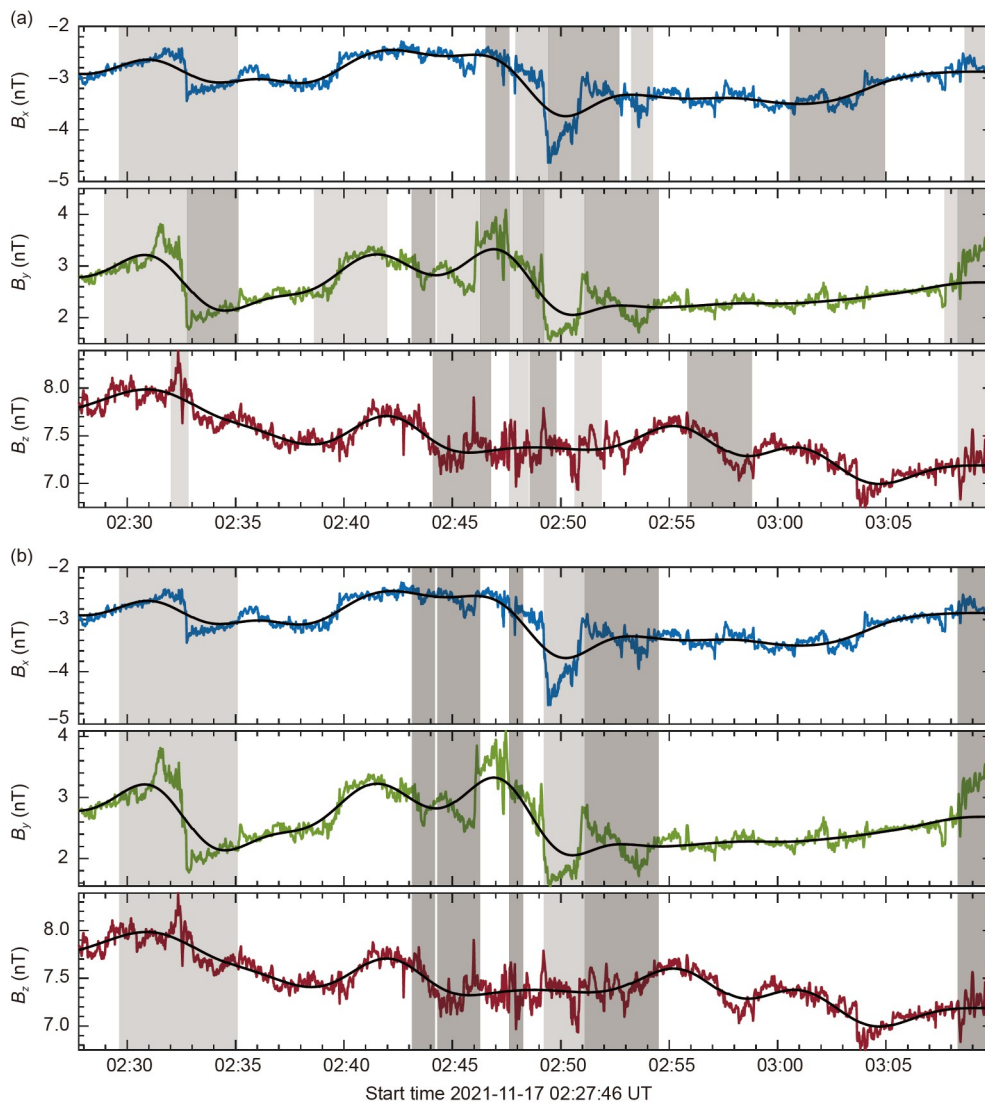
The selection of Alfvénic fluctuation events may affect the precision of the zero offset. Thus, it is a key step to acquire enough highly Alfvénic fluctuation events. In well-calibrated magnetic field data, researchers have some methods such as characterizing an Alfvén wave as a large fluctuation in some components but with the magnitude of total magnetic field being almost constant, or utilizing the high correlation between magnetic field vector and the plasma velocity [12]. However, in the uncalibrated data, we only have the magnetic field without the zero offset corrected. In such a case, an Alfvén wave may not keep a constant magnitude, but the

large rotation in some components should be still there. Therefore, the time intervals with large rotations in some magnetic field components are considered as candidate Alfvénic fluctuations. The selected candidate Alfvénic fluctuations should be long enough in time to suit the Wang-Pan method but not too long to be mixed with compressional structures. Since the Alfvénic fluctuation events always exist in the quiet solar wind, we first manually and roughly select the time intervals without notable structures in the solar wind. Then we search the candidate Alfvénic fluctuation events in three components of magnetic field by choosing the time intervals with large rotation characteristics. We use a 5 s-lowpass Butterworth filter to get rid of high-frequency noise, and further use a 300 s-lowpass Butterworth filter to remove the slow change of the background magnetic field. Then we search the time interval according to the following criteria: a magnetic field component oscillates around the

zero value by at least three times within 0.2–10 min, and the amplitude of the oscillation is greater than 0.6 nT. Any time interval meeting the above criteria is selected to be a candidate.

Figure 2 gives a sample period of Alfvén fluctuation event selection in the spacecraft coordinates from 02:27 to 03:09 UT on 17 November 2021. The selected Alfvénic candidate intervals in each component are indicated by dark-light alternating shadowed regions in Figure 2(a). For each magnetic field component, we can find many candidate Alfvénic intervals. If one candidate Alfvénic interval found in one component overlaps with a candidate interval in another component, we merge them as a single Alfvénic fluctuation event. The start and end time of the component with the larger variance is chosen as the start and end time of the merged Alfvénic event. The merged Alfvénic intervals during this period are shown in Figure 2(b).

The next step is to check if the selected merged intervals



**Figure 2** Examples of finding Alfvén fluctuation events on 17 November 2021 in the spacecraft coordinates. Colored lines are the partially calibrated data and black lines are their ambient magnetic field. (a) The candidate Alfvénic intervals in each magnetic field component in grey shades. (b) The merged candidate Alfvénic fluctuation events are in gray areas, and the dark gray areas are finally used.



could be almost constant in the total magnetic field strength, which is the characteristic of a typical Alfvénic wave, after shifting with an offset. We test the possible offset vector in a range of  $\pm 10$  nT centering on the mean value of each Alfvénic interval with the step of 0.4 nT, i.e., a total of 50 grid points in each component. This process constructs a  $20 \text{ nT} \times 20 \text{ nT} \times 20 \text{ nT}$  arching cube for the zero offset. The variance of the total magnetic field strength, given by  $\frac{\sum_i (B_i - \langle B \rangle)^2}{N}$  (in which  $B_i$  is the magnetic field strength of each data points after a test offset is applied during the interval,  $\langle B \rangle$  is the averaged value of the magnetic field strength during the interval, and  $N$  is the total number of the data points during the interval) is calculated to evaluate how well the criterion of the constant magnitude of an Alfvén wave is satisfied. In our procedure, the variance should less than  $0.012 \text{ nT}^2$ . After testing all the possible offsets, we get the distribution of the variance in the searching cube. All the candidate intervals satisfy the requirements of the variance are treated as the finally selected Alfvénic fluctuation event.

For each finally selected Alfvénic fluctuation event, we investigate all the slices perpendicular to one direction, e.g., to the  $x$ -axis as shown in Figure 3(a), in the search cube and locate the position of the minimum variance in each slice. Then we fit these positions with a linear line as indicated by the blue line in Figure 3(a). We repeat the above procedure for the other two directions, i.e., the  $y$  and  $z$ -axes, and in the three fitting lines, we choose the best fit one as the optimal offset line (OOL) of the Alfvénic fluctuation event. It means that we could get an OOL for each Alfvénic fluctuation event. Then for every 10 adjacent Alfvénic fluctuation events, i.e., for every 10 OOLs, we locate a point, i.e., the position of the most optimal offset, from which the sum of the distances to all the 10 OOLs is the shortest as illustrated by Figure 3(b). Note, since the qualities of the finally selected Alfvénic fluctuation events are different, we assign the

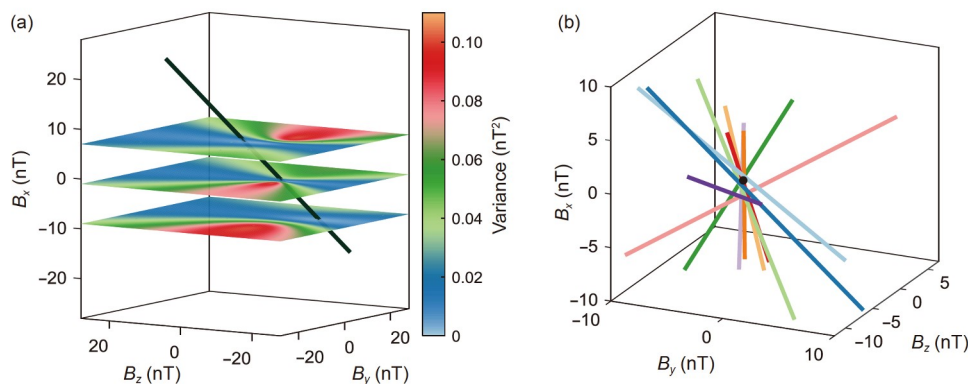
OOLs different weights following the method in the paper by Hu et al. [25], and therefore the distances to the 10 OOLs have different weights when we calculate the sum of the distances.

During the period from November 16 to December 31, 2021, we totally find 824 Alfvénic fluctuation events corresponding to 824 OOLs. Figure 4 displays the obtained offsets for the outer sensor during the period of interest with the orange dots. Considering that the offset should slowly change with time, we smooth and extrapolate the scattered offsets to a relatively smooth curve. We smooth the offset in a 72-h sliding window, which is nearly nine orbit periods, by using the local regression with weighted linear least squares and a 2nd degree polynomial model. Then we use the linear interpolation method to get the offset values throughout the period. It means that the offsets obtained based on the Alfvénic fluctuation events in solar wind are also applied to the magnetosheath region.

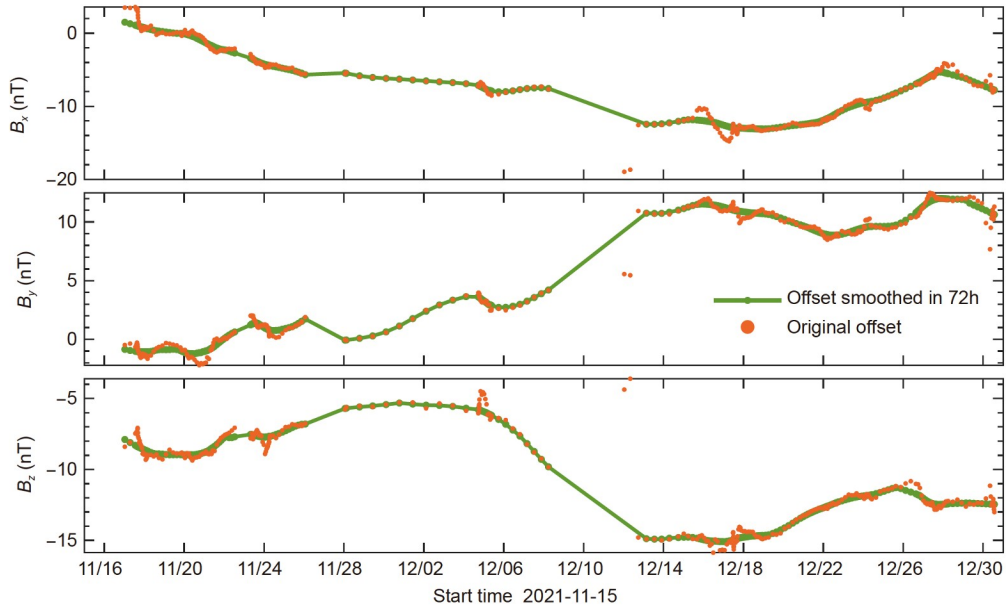
It would be worth emphasizing again that the zero offset contains the quasi-steady interference from the satellite platform and the offset caused by aging of the material of the sensors and energetic particle radiation dose on the sensors. Such effects change the zero offset slowly at the time scale of days to even weeks. Our method determines the zero offset based on Alfvénic fluctuation events at the time scale of hours. Thus, the zero offset could be cleaned effectively.

#### 4 The calibrated data

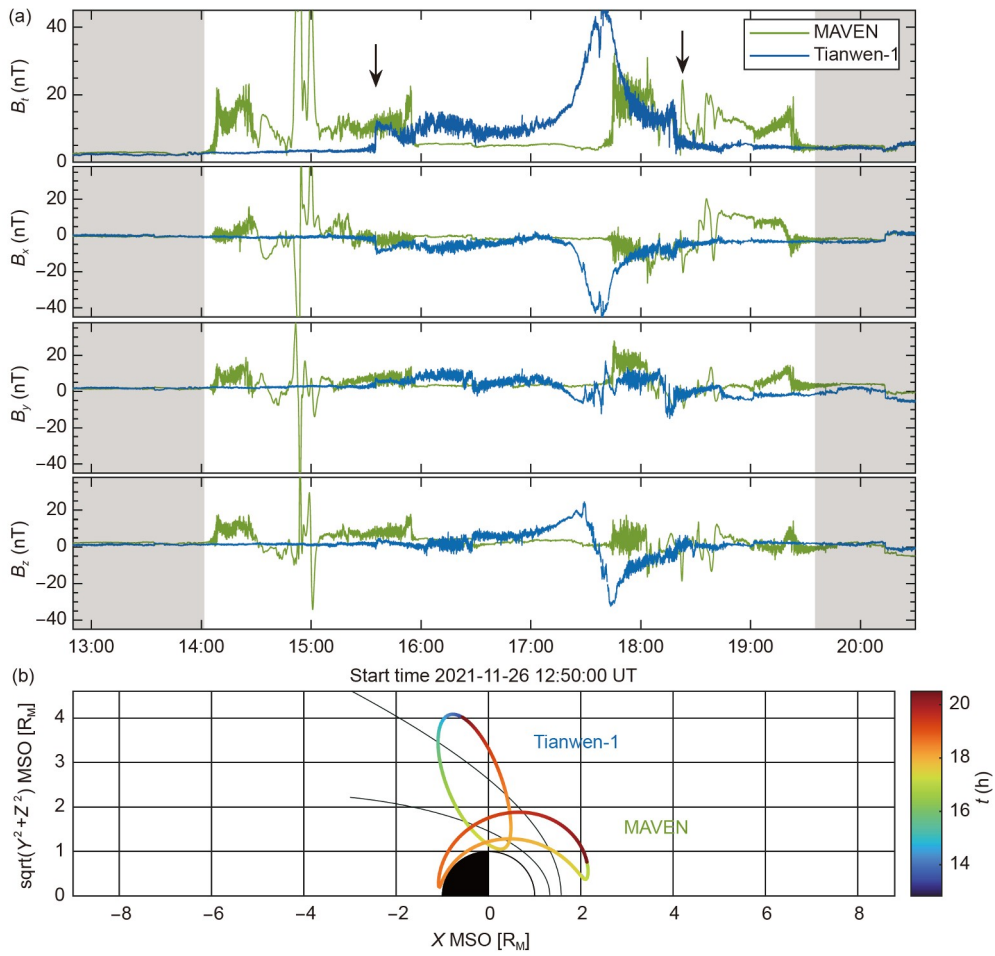
After removing the artificial jumps and offsets, we obtain the scientific Level 2 (or called Level C following China's convention) data of MOMAG by converting the magnetic field vectors from spacecraft coordinates into physical-based coordinates, e.g., the Mars-centered solar orbital (MSO) coordinates. Figure 5(a) shows the processed data between 12:50 and 20:30 UT on 26 November 2021 in blue lines. The



**Figure 3** A diagram of solving offset process. (a) The solution procedure of the OOL. The colored slices in the offset cube represent the variances of an Alfvén interval added the test offsets in the plane. The points with the minimum variance in each slice are fitted into the black line called OOL. (b) A zero offset (black point) formed by ten optimal offset lines shown in colored lines.



**Figure 4** The calculated offset (orange) and the smoothed offset (green) in spacecraft coordinate system during November 16 and December 31, 2021.



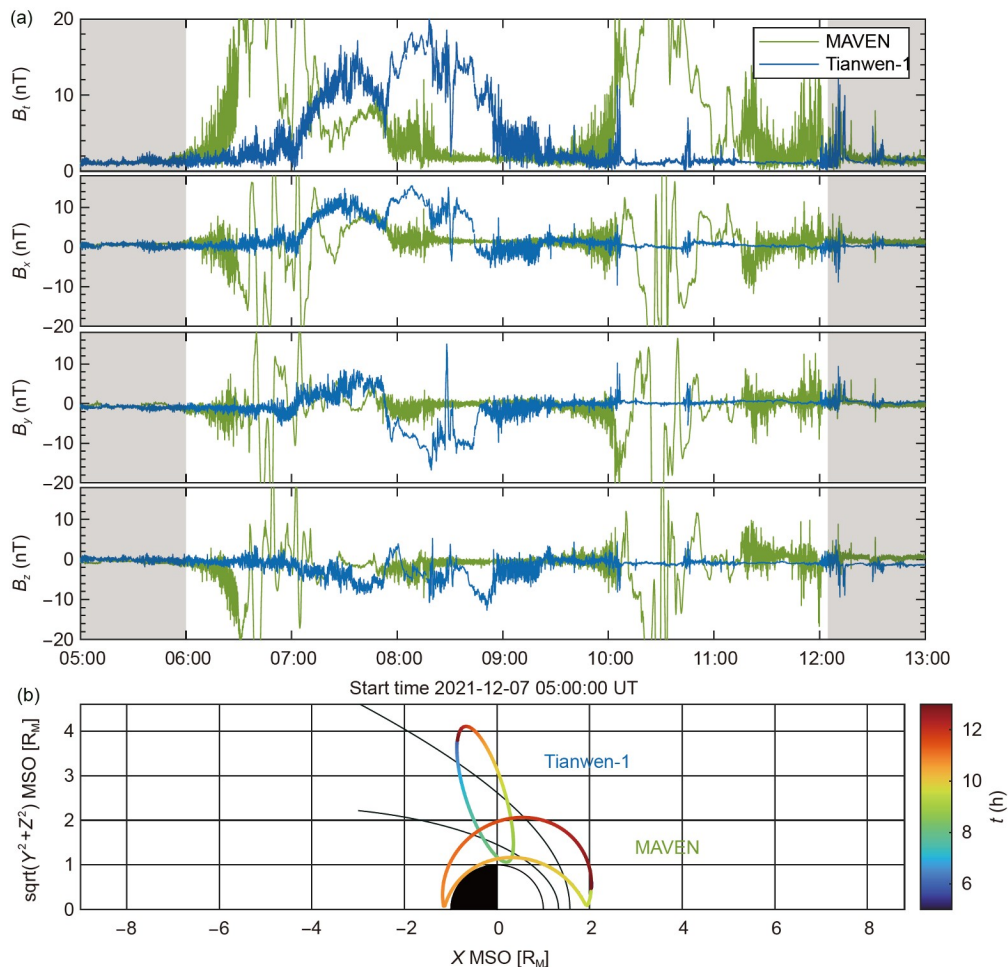
**Figure 5** The scientific data detected by Tianwen-1 and MAVEN between 12:50 and 20:30 UT on 26 November 2021. (a) The processed magnetic field in MSO coordinate system. The arrows indicate the positions of the bow shock. The periods when the two satellites are both in the solar wind are marked in shadows. (b) Their orbit position in colored line.

orbit is shown in Figure 5(b) in the color-coded line. From the figure, we can see that the orbiter crossed the Martian bow shock and entered the magnetosheath at  $\sim 15:34$  UT, and crossed the bow shock again at  $\sim 18:20$  UT when it flew from the magnetosheath to the solar wind.

Thank to MAVEN still working at Mars, we may assess the reliability of the calibrated data by comparing the data with the MAVEN's MAG data [4] shown in green lines in Figure 5(a). The orbital period of Mavem is  $\sim 4$  h and it came into the magnetosphere twice during the period of interest. We could compare the magnetic field detected by them when they were both in the solar wind, which are denoted by shadows. The magnetic field they detected is similar in the solar wind. The magnetic field they detected is similar in the solar wind. The mean magnetic field of Tianwen-1 in the solar wind in the two shadows in  $x$ ,  $y$ ,  $z$  directions in MSO coordinate system were  $-0.6$ ,  $0.5$ , and  $0.9$  nT respectively. In MAVEN's observation, they were  $-0.6$ ,  $1.7$ , and  $0.9$  nT. The mean magnitudes were  $3.2$  nT in Tianwen-1 and  $3.6$  nT in MAVEN. At  $\sim 20:14$  UT both the spacecraft observed a sudden change. In the  $x$  component, the magnetic field at both spacecraft jumped from negative to positive, while in the other two components, the magnetic field from positive to negative

with slightly difference in magnitude. We select 5 min data around the sudden change, use the correlation analysis method, and obtain the shifting time is  $\sim 14$  s with the Tianwen-1 lagging behind. Tianwen-1 orbiter has a quite different orbit from MAVEN as illustrated in Figure 5(b). The distance between Tianwen-1 orbiter and MAVEN was more than thousands of kilometers at that time. Thus, it is natural that there are differences between the data from the two spacecrafts.

Figure 6 shows the processed data and the corresponding MAVEN magnetic field between 05:00 and 13:00 UT on 07 December 2021 in the same format as Figure 5. The magnitude and components of their magnetic field are similar in the solar wind. The mean magnetic field of Tianwen-1 in  $x$ ,  $y$ ,  $z$  directions in the solar wind were  $0.4$ ,  $-0.4$ , and  $-0.6$  nT, respectively. In MAVEN observation, they were  $0.9$ ,  $-0.4$ , and  $-0.2$  nT. Both of the mean magnitudes were  $1.3$  nT. Different from the bow shocks in Figure 5, the signature of the bow shock observed by Tianwen-1 and MAVEN is not clear because of the upstream solar wind conditions. Besides, around 12:31 UT, both MAVEN and Tianwen-1 observed a set of strong fluctuations of the magnetic field.

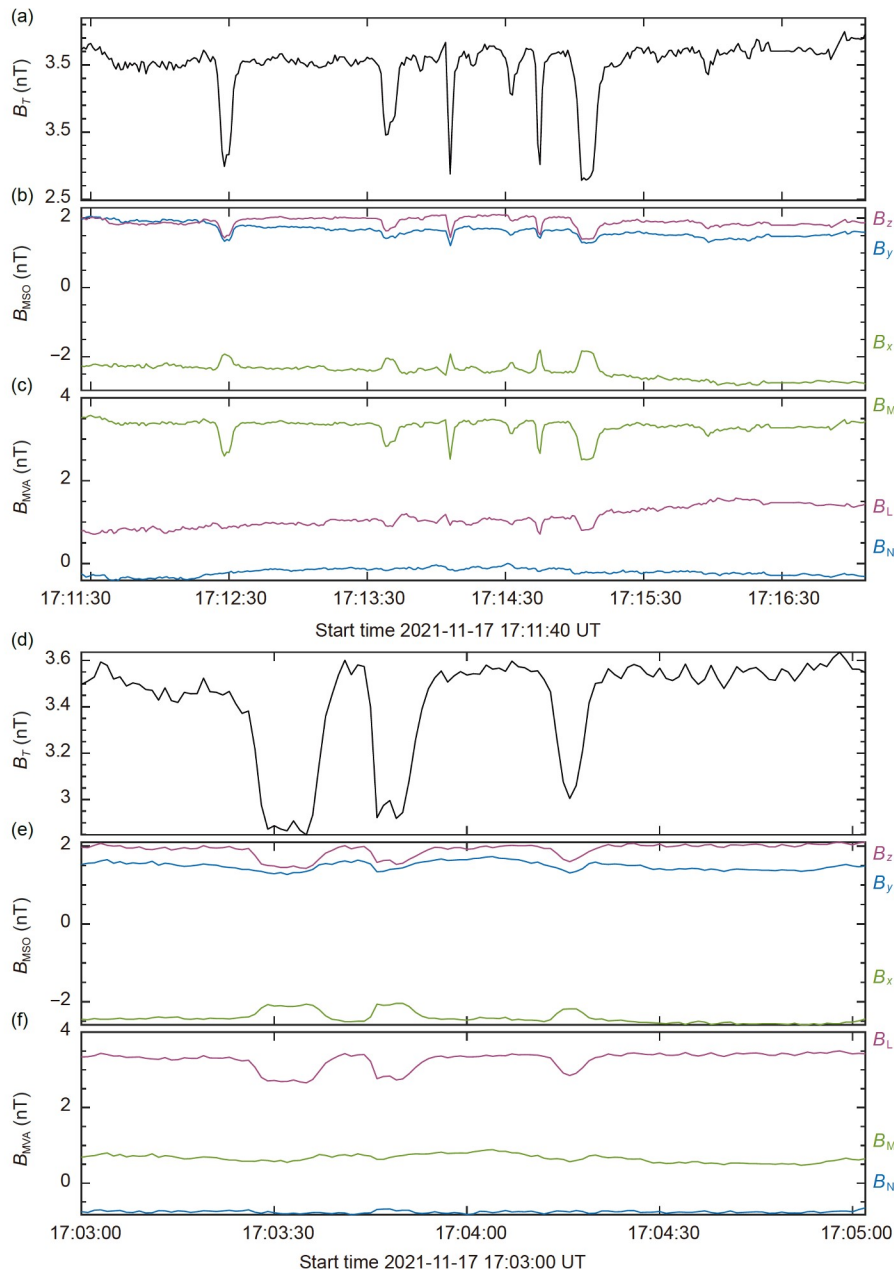


**Figure 6** The scientific data detected by Tianwen-1 and MAVEN between 05:00 and 13:00 UT on 07 December 2021 in the same format as Figure 5.

Except the featured structures we compared in the above two episodes, we also investigate some magnetic holes (MHs) that are widely detected in the space environments [26], especially in the solar winds [27,28]. Some studies have detected the MHs in the solar wind surrounding Mars [29]. The typical characteristic of a MH is that the magnetic field amplitude has a large dip. An example is given in Figure 7. There was a serial of MHs in just 5 min on 17 November 2021. The magnetic field magnitudes shown in Figure 7(a) have some large dips, which are corresponding to the MHs. Figure 7(b) shows the three magnetic field components in

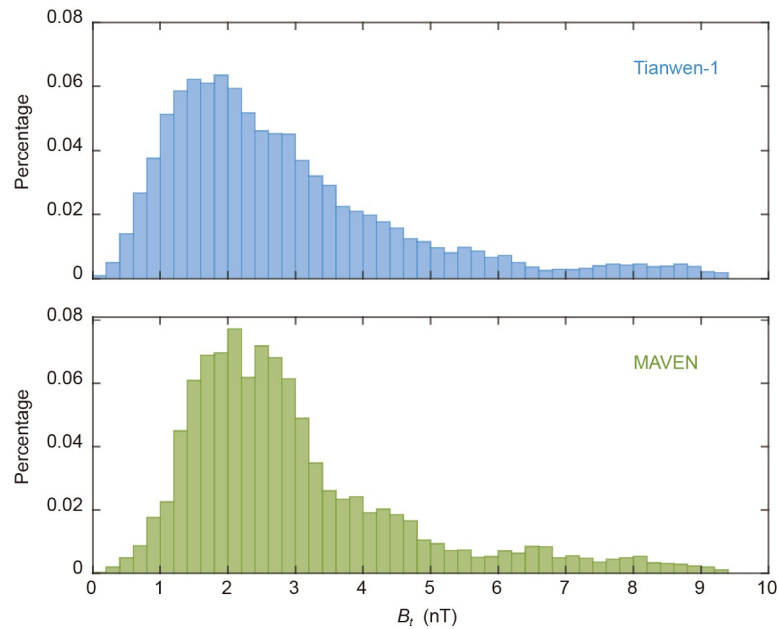
MSO coordinates. The directions of their magnetic field changed slowly, and therefore they could be called linear MHs.

We rotate the magnetic field to the LMN coordinate system shown in Figure 7(c) using the MVA method. Because we apply the MVA method in over 5 min duration rather than the duration of only one MH, the magnetic field in the maximum variation directions ( $B_L$ ) denotes the slow increasing of the ambient magnetic field. The characteristic of large dips appears in the intermediate variation directions (M), and the ambient magnetic field is mainly in the L direction.



**Figure 7** The serial magnetic holes observed in Tianwen-1 and MAVEN on 17 November 2021. (a) The magnetic field magnitude of Tianwen-1. (b) Three magnetic field components of Tianwen-1 in MSO coordinates. (c) Three magnetic field components of Tianwen-1 in LMN coordinates. (d) The magnetic field magnitude of MAVEN. (e) Three magnetic field components of MAVEN in MSO coordinates. (f) Three magnetic field components of MAVEN in LMN coordinates.





**Figure 8** The distributions of the magnetic field strength in the solar wind based on Tianwen-1/MOMAG and MAVEN during November 16 and December 31, 2021.

We also find a series of MH structures around 17:04 UT in MAVEN/MAG data shown in Figure 7(d)–(f). Their dips are smaller than the MHs observed by the Tianwen-1. They appeared earlier at MAVEN than at Tianwen-1, since MAVEN was at the upstream of Tianwen-1. Whether they are mirror modes and their creation environment require further confirmation.

Alternatively, we can statistically compare the distributions of the total magnetic field strength in the solar wind from the two spacecraft as shown in Figure 8. Based on the data during November 16 and December 31, 2021, there are about 314 h MAVEN data and 400 h Tianwen-1 data in the solar wind, and about 117 h when both MAVEN and Tianwen-1's orbiter stayed in the solar wind. Based on the 117 h data, it could be found that the magnetic field magnitude distribution from MOMAG is similar to that from MAVEN/MAG. The mean value is 3.02 nT for Tianwen-1 and 3.07 nT for MAVEN. The deviation of the mean values is only 0.05 nT, which is negligible compared to the magnetic field strength in the solar wind. The comparison suggests that the measured magnetic fields by the two spacecraft are quite consistent, and our calibrated data are credible. Liu et al. [30] has made a detailed statistical survey of IMF based on MAVEN data.

## 5 Conclusions

In this work, we present the process of the in-flight calibration of the Tianwen-1/MOMAG magnetic field data as well as the data quality. We develop a series of procedures to

clean the interference caused by the spacecraft. The dynamic fields, i.e., the artificial jumps, are cleaned firstly based on the different jump magnitudes at the dual sensors. Then we use properties of Alfvénic waves to correct the offset of measured magnetic field. It should be noted that the errors produced by the misalignment between of the sensors' sensitive axes and spacecraft reference axes have been calibrated and assessed at the ground.

After the in-flight calibration, one and a half months' calibrated magnetic field data are analyzed and compared with MAVEN/MAG data. The common typical structures, such as physical jumps and MHs, could be found in the data from the both spacecrafts. The distribution of magnetic field strength in the solar wind measured by Tianwen-1/MOMAG is similar to that by MAVEN/MAG with only 0.09 nT deviation in the median value. These results suggest that the difference of the data from the two spacecraft could be neglected and the calibrated data are reliable.

*This work was supported by the National Natural Science Foundation of China (Grant Nos. 42130204, 42188101 & 42241155) and the Strategic Priority Program of the Chinese Academy of Sciences (Grant No. XDB41000000). Y.W. is particularly grateful to the support of the Tencent Foundation. We acknowledge the use of the data from the MAG and SWI onboard MAVEN spacecraft, which are obtained from NASA Planetary Data System (<https://pds-ppi.igpp.ucla.edu/>). The Tianwen-1/MOMAG data are publicly available at CNSA Data Release System (<http://202.106.152.98:8081/marsdata/>) or the data used in this paper can just be downloaded from the official website of the MOMAG team ([http://space.usc.edu.cn/dreams/tw1\\_momag/](http://space.usc.edu.cn/dreams/tw1_momag/)).*

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